AIRCRAFT SYSTEMS

Course Code		Category	Hours / Week			Credits	Maximum Marks		
AAEB21		Core	L	Т	Р	C 3	CIA	SEE	Tota
			3	-	-		30	70	100
Contact Classes: 45		Tutorial Classes: Nil	Practical Class			es: Nil	Total Classes: 45		es: 45
OBJECT The cour		ble the students to:							
Ι	The basic obj	ects oriented programmin	ng conce	epts and	l apply	them in pro	blem sol	ving.	
II	The fundamental concepts of computer programming, structure of the computational process and the basic approaches to the design of software applications.								
III	The implement	tation of file processing, inte	er proces	ss comm	unicatio	on and excep	tion handl	ing techn	iques
IV	To design and	develop Graphical user inter	rface usi	ing Appl	ets and	Swing/AW7	^c concepts		
	E OUTCOME								
After su	ccessful comp	letion of the course, stud	dents w	vill be a	ble to:				
CO 1		the concept of aircraft sys			•			•	
	•	ns, and sub-systems for un		<u> </u>		• •		•	
CO 2									
	systems for better design and operation of aircraft systems.								
CO 3	Make use of electrical power generation and control systems of Power distribution in primary								
GO 4	and secondary control for Power conversion and energy storage								
CO 4	Select the basic air cycle and vapour cycle systems for obtaining additional cooling effect								
CO 5	through evaporation. Illustrate the principal operation of hydraulic system its function, merits, application, design								
CO 5	requirements and Hydraulic fluid properties for transforming the energy in landing gear.								
CO 6	Explain the working principles of pneumatic system and break management system in landi							nding	
000	-	and easy operation.	inacio s	y scorri c		un munuger	iiciii: 5950		lang
CO 7	Demonstrate the working principle of aircraft gas turbine engines its airframe interfaces for								
		air and fuel flow in aircra							
CO 8	Interpret the need of Fuel systems, its characteristics, safety and fuel inserting system for								ſ
	ensuring unin	terrupted flow in the engi	ine						
CO 9	Recall the methods of flight, engine and modern control systems used in aircraft for safe and								
CO 10	sustained fligh					for only on sin	6		
CO 10		uturistic application of mo ol systems and autopilot s		ontrol s	ystem I	for enhancin	ig aircrai	t periorn	nance,
MODUI	LE -I INTRO	DDUCTION TO AIRCE	RAFT S	SYSTE	MS			Classes	: 10
Aircraft	systems- airfra	systems; Generic system of ame systems, vehicle systems, of requirements, mission	ems, av	vionics s	systems	s, mission s	ystems ar		

ELECTRICAL SYSTEMS AND AIR CONDITIONING, **MODULE -II** Classes: 10 PRESSURIZING SYSTEMS Electrical loads in aircraft. Electrical power generation and control- DC, AC- types. Power distributionprimary, secondary. Power conversion and energy storage; Load protection; Electrical load management systems, 270 V DC systems; Basic air cycle systems; Vapour cycle systems, boost-strap air cycle system; Evaporative Vapour cycle systems; Evaporative air cycle systems; Oxygen systems; deicing and anti-icing systems MODULE -III HYDRAULIC SYSTEMS AND PNEUMATIC SYSTEMS Classes: 08 Hydraulic systems: function, merits, application, system loads, design requirements; Principal components; Hydraulic fluid: required properties; Hydraulic piping, pumps, reservoir, accumulator; Pneumatic systems ; Advantages:- Working principles ; Typical air pressure system ; Brake system; Typical pneumatic power system; Components, landing gear systems; Landing gear and brake management systems. MODULE -IV ENGINE CONTROL AND FUEL SYSTEMS Classes: 08 Principle of operation of aircraft gas turbine engines; Engine - airframe interfaces; Control of fuel flow, air flow, Limited authority control systems, full authority control systems- examples; Power off takesneed, types; Fuel systems- characteristics, components, operating modes; Fuel tank safety- fuel inserting system MODULE-V AIRPLANE CONTROL SYSTEMS Classes: 08 Flight control systems- primary and secondary flight control conventional systems; Power assisted and fully powered flight controls ; Power actuated systems; Engine control systems; Push pull rod system, flexible push full rod system; Control linkages, actuation- types, description and redundancy. Components; Modern control systems; Digital fly by wire systems, control laws, implementation; Auto pilot system. **Text Books:** 1. Moir, I.andSeabridge, A, —AircraftSystems: Mechanical, Electrical and AvionicsSubsystems Integration I, John Wiley, 3rd Edition2008. 2. Moir, I. and Sea bridge, A, -Design and Development of Aircraft Systems- An Introduction, AIAA Education Series, AIAA, 2004. **Reference Books:** 1. Pallett, E.H.J., —Aircraft Instruments and Integrated Systems^{II}, Longman Scientific &Technical 10th edition.1992. 2. Harris, D, —Flight Instruments and Automatic Flight Control Systems^I, 6th edition,2004. 3. Bolton, W., "Pneumatic and Hydraulic Systems", Butterworth-Heinemann. Web References 1.https://www.aircraftsystemscomjet.com/ 2.https://www.srmuniv.ac.in/sites/default/files/downloads/Aircraft ctrl Systems.pdf 3.https://hydraulicspneumatics.com/other-technologies/chapter-5-pneumatic-and-hydraulic-systems 4. https://www.stahl.de/fileadmin/Dateien/download publikationen/web havc and pressurization.pdf **E-Text Books:** 1. https://www.amazon.in/Aircraft-Systems-Mechanical-ElectricalIntegration/dp/0470059966 2. https://www.scribd.com/book/142412367/Aircraft-Systems-Mechanical-Electrical-and Avionics