

## INSTITUTE OF AERONAUTICAL ENGINEERING (Autonomous)

Dundigal, Hyderabad - 500043, Telangana

## AERONAUTICAL ENGINEERING

## ATTAINMENT OF COURSE OUTCOME - ACTION TAKEN REPORT

| Name of the faculty: | Mr. I SEETHA RAMA RAO | B             | Aeronautical Engineering |  |
|----------------------|-----------------------|---------------|--------------------------|--|
|                      | THE RANGE (ACC        | Department:   |                          |  |
| Regulation:          | IARE - R18            | Batch:        | 2019-2023                |  |
| Course Name:         | Flight Vehicle Design | Course Code:  | AAEB24                   |  |
| Semester:            | VII                   | Target Value: | 60% (1.8)                |  |

## Attainment of COs:

|     | Course Outcome   | Direct<br>Attainment | Indirect<br>Attainment | Overall<br>Attainment | Observation |
|-----|--|----------------------|------------------------|-----------------------|-------------|
| CO1 | Choose data collection for conceptual sketch from existingaircraft for understanding aerodynamic   | 3.00                 | 2.20                   | 2,8                   | Attained    |
| CO2 | Classify rubber engine sizing of a given fighter aircraft for calculating the take -off weights in order so that the aircraft meets all set requirements.    | 3.00                 | , 2.30                 | 2.9                   | Attained    |
| CO3 | Make use of airfoil geometry and co-ordinates for obtaining the required 3D model by using designer tools like catiaV5.                                      | 3.00                 | 2.20                   | 2.8                   | Attained    |
| CO4 | Simplify the performance estimations involving design layout for calculating the variation of CL and CD at angle of attack.                                  | 2.30                 | 2.20                   | 2.3                   | Attained    |
| CO5 | Estimate take-off gross weight of simple cruise mission profile for calculating the empty weight fraction.   | 3.00                 | 2.30                   | 2.9                   | Attained    |
| CO6 | Identify the total drags on an aircraft and calculate the total weight, thrustand drag for exit pressure and Mach number for the given nozzle configurations | 2.30                 | 2.30                   | 2.3                   | Attained    |

Action Taken Report: (To be filled by the concerned faculty / course coordinator)

H Head of the Department ment

Aeronautical Engineering
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