# **INSTITUTE OF AERONAUTICAL ENGINEERING**

(Autonomous) Dundigal-500043, Hyderabad

B.Tech VII SEMESTER END EXAMINATIONS (REGULAR) - DECEMBER 2023

Regulation: UG-20

TECHNIQUES IN WIND TUNNEL TESTING

Time: 3 Hours

(AERONAUTICAL ENGINEERING)

Max Marks: 70

Answer ALL questions in Module I and II Answer ONE out of two questions in Modules III, IV and V All Questions Carry Equal Marks All parts of the question must be answered in one place only

## $\mathbf{MODULE}-\mathbf{I}$

- 1. (a) What do you understand by the wind tunnel? Explain the construction and working of a subsonic wind tunnel with a neat sketch. [BL: Understand] CO: 1|Marks: 7]
  - (b) A student team is to design a human-powered submarine for a design competition. The overall length of the prototype submarine is 4.85 m, and its student designers hope that it can travel fully submerged through water at 0.440 m/s. The water is freshwater (a lake) at  $T = 15^{0}C$ . The design team builds a one-fifth scale model to test in their university's wind tunnel . A shield surrounds the drag balance strut so that the aerodynamic drag of the strut itself does not influence the measured drag. The air in the wind tunnel is at  $T = 25^{0}C$  and at one standard atmosphere pressure. At what air speed do they need to run the wind tunnel in order to achieve similarity? Comment on your result.

For water at  $T = 15^{\circ}C$  and atmospheric pressure,  $\rho = 999.1 kg/m^3$  and  $\mu = 1.138 * 10^{-3} kg/ms$ . For air at  $T = 25^{\circ}C$  and atmospheric pressure,  $\rho = 1.184 kg/m^3$  and  $\mu = 1.849 * 10^{-5} kg/ms$ . [BL: Apply] CO: 1|Marks: 7]

#### $\mathbf{MODULE}-\mathbf{II}$

- 2. (a) Write short notes on the following:
  - i) Power losses
  - ii) Wind tunnel correction factor
  - iii) Tunnel blockage factor [BL: Understand | CO: 2|Marks: 7]
  - (b) Summarize various losses in wind tunnel. Discuss the impact of test section flow quality on wind tunnel performance. [BL: Understand| CO: 2|Marks: 7]

### $\mathbf{MODULE}-\mathbf{III}$

3. (a) Briefly explain how force measurements are carried out using an external strain gauge balance.

[BL: Understand| CO: 3|Marks: 7]

(b) Enumerate the key support points in a three-point wire support system for aerodynamic models, and how do they help to maintain the model's position within the wind tunnel test section?

[BL: Understand| CO: 3|Marks: 7]

4. (a) Discuss briefly on types of external balances for load measurements in wind tunnel with neat sketches. [BL: Understand| CO: 4|Marks: 7]

(b) How does a platform balance address interference with airflow around the model, and what design considerations are taken into account to minimize such interference?

[BL: Understand |CO: 4 |Marks: 7]

### $\mathbf{MODULE}-\mathbf{IV}$

- 5. (a) Describe the techniques and devices used for measuring steady and unsteady pressure in wind tunnels? [BL: Understand] CO: 5|Marks: 7]
  - (b) Write in detail about the principle and working of Laser Doppler Anemometer (LDA). How LDA principle is used to measure the velocity in a wind tunnel. [BL: Understand] CO: 5[Marks: 7]
- 6. (a) What are the primary temperature measurement techniques employed in wind tunnel experiments, and how do they differ in terms of accuracy and applicability across various test scenarios? [BL: Understand] CO: 5|Marks: 7]
  - (b) Explain the calibration techniques of wind tunnels. How calibration of supersonic wind tunnel is different from subsonic wind tunnel? [BL: Understand] CO: 5|Marks: 7]

#### $\mathbf{MODULE}-\mathbf{V}$

7. (a) List the optical methods used for flow visualization. Explain the shadowgraph system flow visualization techniques in detail with the help of neat sketch.

[BL: Understand] CO: 6|Marks: 7]

- (b) Summarize the interferometer flow visualization in detail with a neat sketch. Identify the basic flow variable is measured using this technique. [BL: Understand] CO: 6[Marks: 7]
- 8. (a) Discuss the use of tufts and electrical techniques for flow visualization studies.

[BL: Understand CO: 6 Marks: 7]

(b) Illustrate the phenomenon of separation of flow over a 2D wing with the help of liquid paraffin generated smoke wire technique with good sketches. What are its merits over kerosene generated smoke? [BL: Apply] CO: 6|Marks: 7]

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