

COMPUTATIONAL STRUCTURAL ANALYSIS LABORATORY

VII Semester: AE																																				
Course Code	Category	Hours / Week			Credits	Maximum Marks																														
AAE111	Core	L	T	P	C	CIE	SEE	Total																												
		-	-	3	2	30	70	100																												
Contact Classes: Nil		Tutorial Classes: Nil		Practical Classes: 36			Total Classes: 36																													
<p>I. COURSE OVERVIEW: Computational Structural Analysis Laboratory sessions focus on the creation of geometry, meshing (Discretization) and the physics behind the stress strain variation on a continuum. It will also cover the different solvers available in a FEA package and their applications based on the problem type. This course offers a wide range of applications in aircraft structural analysis such as deflection of truss, frames, beams, stress and strain distributions in a plate as well as a solid continuum. Apart from these, it will also address the nonlinear stress problems alongside vibration and flutter analysis.</p> <p>II. OBJECTIVES: The course should enable the students to: I. Make the student familiar with latest computational techniques and software used for structural analysis. II. Enable the student get a feeling of how real-life structures behavior for static and dynamics loads. III. Become familiar with professional and contemporary issues in the design and fabrication.</p> <p>III. COURSE OUTCOMES: After successful completion of the course, students should be able to:</p> <table style="width: 100%; border: none;"> <tr> <td style="width: 15%;">CO 1</td> <td style="width: 65%;">Explain the computational methods and Software's that are used in aerospace fields to simulate the complex problems through ANSYS.</td> <td style="width: 20%; text-align: right;">Understand</td> </tr> <tr> <td>CO 2</td> <td>Solve the parameters like deflections, stress, strain and bending moment by using ANSYS for the linear and non-linear problems that occur in aircraft structural components (beams, bars etc.).</td> <td style="text-align: right;">Apply</td> </tr> <tr> <td>CO 3</td> <td>Calculate the numerical solution of static structural problems using discretization methods and convergence criteria to minimize the errors.</td> <td style="text-align: right;">Analyze</td> </tr> <tr> <td>CO 4</td> <td>Select the appropriate heat transfer mechanism using ANSYS thermal workbench for efficient cooling of on board avionics system.</td> <td style="text-align: right;">Analyze</td> </tr> <tr> <td>CO 5</td> <td>Predict the suitable appropriate results using governing equations for vibration problems that occur in aircraft structural components (beams, spring-mass system)</td> <td style="text-align: right;">Evaluate</td> </tr> <tr> <td>CO 6</td> <td>Determine the nature of stress-strain distribution by using appropriate governing equations for an aircraft structural components such as wings, fuselage and landing gear.</td> <td style="text-align: right;">Evaluate</td> </tr> </table> <p>IV. SYLLABUS:</p> <table style="width: 100%; border: none;"> <tr> <th colspan="2" style="text-align: center; padding: 5px;">LIST OF EXPERIMENTS</th> </tr> <tr> <td style="width: 15%; padding: 5px;">Week-1</td> <td style="padding: 5px;">INTRODUCTION AND BASIC FUCTIONS</td> </tr> <tr> <td colspan="2" style="padding: 5px;"> a. Starting up of ANSYS/Nastran b. Description of user interface </td> </tr> <tr> <td style="padding: 5px;">Week-2</td> <td style="padding: 5px;">STATIC ANALYSIS: TRUSS AND FRAME STRUCTURES</td> </tr> <tr> <td colspan="2" style="padding: 5px;"> a. 2-D truss structures b. 3-D truss structures </td> </tr> </table>									CO 1	Explain the computational methods and Software's that are used in aerospace fields to simulate the complex problems through ANSYS.	Understand	CO 2	Solve the parameters like deflections, stress, strain and bending moment by using ANSYS for the linear and non-linear problems that occur in aircraft structural components (beams, bars etc.).	Apply	CO 3	Calculate the numerical solution of static structural problems using discretization methods and convergence criteria to minimize the errors.	Analyze	CO 4	Select the appropriate heat transfer mechanism using ANSYS thermal workbench for efficient cooling of on board avionics system.	Analyze	CO 5	Predict the suitable appropriate results using governing equations for vibration problems that occur in aircraft structural components (beams, spring-mass system)	Evaluate	CO 6	Determine the nature of stress-strain distribution by using appropriate governing equations for an aircraft structural components such as wings, fuselage and landing gear.	Evaluate	LIST OF EXPERIMENTS		Week-1	INTRODUCTION AND BASIC FUCTIONS	a. Starting up of ANSYS/Nastran b. Description of user interface		Week-2	STATIC ANALYSIS: TRUSS AND FRAME STRUCTURES	a. 2-D truss structures b. 3-D truss structures	
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Week-3	STATIC ANALYSIS: BEAMS
<ul style="list-style-type: none"> a. Straight beams b. Tapered beams 	
Week-4	STATIC ANALYSIS: TWO DIMENSIONAL PROBLEMS
<ul style="list-style-type: none"> a. 2-D structure with various loadings b. 2-D structures with different materials c. Plate with hole 	
Week-5	DYNAMIC ANALYSIS: MODAL AND TRANSIENT ANALYSES
<ul style="list-style-type: none"> a. Modal analysis b. Transient Response (spring-mass system) 	
Week-6	THERMAL ANALYSIS
<ul style="list-style-type: none"> a. Bars and beams b. 2D structures 	
Week-7	NON LINEAR ANALYSIS
<ul style="list-style-type: none"> a. Nonlinear behavior (Large deflections) b. Nonlinear behavior (Materials) 	
Week-8	HARMONIC RESPONSE ANALYSIS
<ul style="list-style-type: none"> a. Random Vibration Analysis of a Deep Simply-Supported Beam b. Harmonic Response of a Spring-Mass System 	
Week-9	ANALYSIS OF AIRCARFT STRUCTURE: WING
<ul style="list-style-type: none"> a. Static analysis of Aircraft wing structure b. Modal analysis of aircraft wing structure 	
Week-10	ANALYSIS OF AIRCARFT STRUCTURE:FUSELAGE
<ul style="list-style-type: none"> a. Static analysis of Aircraft Semi monoque fuselage structure b. Modal analysis of aircraft Semi monoque fuselage structure 	
Week-11	ANALYSIS OF AIRCARFT STRUCTURE:LANDING GEAR
<ul style="list-style-type: none"> a. Static analysis of main landing gear b. Modal analysis of main landing gear 	
Week-12	ANALYSIS OF COMPOSITE STRUCTURES
<ul style="list-style-type: none"> a. Static analysis of composite bar and beam b. Static analysis of composite plate 	
Reference Books:	
<ol style="list-style-type: none"> 1. Huei-Huang Lee, “Finite Element Simulations with ANSYS Workbench 16”, SDC publications, 2nd Edition, 2016. 2. Anderson, William J “MSC/Nastran: Interactive Training Program” Wiley 1st Edition 2015. 	
Web Reference:	
http://www.iare.ac.in	

[Course Home Page:](#)

SOFTWARE AND HARDWARE REQUIREMENTS FOR A BATCH OF 36 STUDENTS:

SOFTWARE: ANSYS 16 or MSC Nastran

HARDWARE: Desktop Computers with 4 GB RAM 36 nos