

## FLIGHT VEHICLE DESIGN LABORATORY

| <b>VII Semester: AE</b>   |   |                              |   |                              |         |               |                          |       |
|---|---|------------------------------|---|------------------------------|---------|---------------|--------------------------|-------|
| Course Code   | Category  | Hours / Week                 |   |                              | Credits | Maximum Marks |                          |       |
| AAE112  | Core  | L                            | T | P                            | C       | CIA           | SEE                      | Total |
|   |   | -                            | - | 3                            | 2       | 30            | 70                       | 100   |
| <b>Contact Classes: Nil</b>   |   | <b>Tutorial Classes: Nil</b> |   | <b>Practical Classes: 36</b> |         |               | <b>Total Classes: 36</b> |       |
| <b>I. COURSE OVERVIEW:</b>  |   |                              |   |                              |         |               |                          |       |
| <p>The aim of Flight Vehicle design (FVD) LAB is to introduce students the overview of the design process. The course covers basic principles of conceptual design process of an aircraft and the related details of all design techniques. After completion of the course the student gains adequate knowledge to design all the different phase of an aircraft design. Weight estimation for different aircrafts</p>  |   |                              |   |                              |         |               |                          |       |
| <b>II. OBJECTIVES:</b>  |   |                              |   |                              |         |               |                          |       |
| <b>The course should enable the students to:</b>  |   |                              |   |                              |         |               |                          |       |
| <ul style="list-style-type: none"> <li>I Understand the basic skills involved in weight estimation for aircraft conceptual design process.</li> <li>II Illustrate relevant theoretical knowledge, applicable for initial sizing and configuration layout of aircraft.</li> <li>III Evaluate basic techniques for design of aircraft using given design requirement and mission profiles.</li> </ul>   |   |                              |   |                              |         |               |                          |       |
| <b>III. COURSE OUTCOMES:</b>  |   |                              |   |                              |         |               |                          |       |
| <b>After successful completion of the course, students should be able to:</b>   |   |                              |   |                              |         |               |                          |       |
| <ul style="list-style-type: none"> <li>CO 1 <b>Choose</b> data collection for conceptual sketch from existing aircraft for understanding aerodynamic performance requirements. <span style="float: right;">Apply</span></li> <li>CO 2 <b>Classify</b> rubber engine sizing of a given fighter aircraft for calculating the take-off weights in order so that the aircraft meets all set requirements <span style="float: right;">Analyze</span></li> <li>CO 3 <b>Make</b> use of airfoil geometry and co-ordinates for obtaining the required 3D model by using designer tools like catiaV5. <span style="float: right;">Apply</span></li> <li>CO 4 <b>Simplify</b> the performance estimations involving design layout for calculating the variation of C L and CD at angle of attack. <span style="float: right;">Analyze</span></li> <li>CO 5 <b>Estimate</b> take-off gross weight of simple cruise mission profile for calculating the empty weight fraction. <span style="float: right;">Evaluate</span></li> <li>CO 6 <b>Identify</b> the total drags on an aircraft and calculate the total weight, thrust and drag for exit pressure and Mach number for the given nozzle configurations <span style="float: right;">Apply</span></li> </ul> |   |                              |   |                              |         |               |                          |       |
| <b>IV. SYLLABUS:</b>  |   |                              |   |                              |         |               |                          |       |
| <b>LIST OF EXPERIMENTS</b>  |   |                              |   |                              |         |               |                          |       |
| <b>Week-1</b>   | <b>OBJECTIVES AND REQUIREMENTS OF THE VEHICLE</b> |                              |   |                              |         |               |                          |       |
| <p>Data collection for conceptual sketch from existing aircraft includes :</p> <ul style="list-style-type: none"> <li>a. Type, Role, Mission.</li> <li>b. Payload</li> <li>c. Aerodynamic &amp; performance requirements.</li> </ul>  |   |                              |   |                              |         |               |                          |       |
| <b>Week-2</b>   | <b>CONCEPTUAL SKETCH AND WEIGHT ESTIMATION</b>    |                              |   |                              |         |               |                          |       |
| <ul style="list-style-type: none"> <li>a. Conceptual sketch of candidate aircraft (3-view).</li> <li>b. First estimation of gross take-off weight with trade-off studies.</li> </ul>  |   |                              |   |                              |         |               |                          |       |

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| <b>Week-3</b>  | <b>AIRFOIL DESIGN AND CONSTRAINT ANALYSIS</b> |
| a. Airfoil and wing geometry selection   |   |
| <b>Week-4</b>  | <b>CONSTRAINT ANALYSIS</b>                    |
| a. Determination of Thrust-to-Weight ratio and Wing Loading                                |   |
| <b>Week-5</b>  | <b>INITIAL SIZING-I</b>                       |
| a. Rubber engine & fixed engine sizing.  |   |
| <b>Week-6</b>  | <b>INITIAL SIZING-II</b>                      |
| a. Configuration layout, crew station, passengers and payload                              |   |
| <b>Week-7</b>  | <b>PERFORMANCE ESTIMATIONS</b>                |
| a. Performance constraint analysis   |   |
| <b>Week-8</b>  | <b>LOAD ESTIMATIONS-I</b>                     |
| a. Landing gear loads  |   |
| <b>Week-9</b>  | <b>LOAD ESTIMATIONS-II</b>                    |
| a. Propulsion system load.   |   |
| <b>Week-10</b>   | <b>COST ESTIMATION</b>                        |
| a. Cost estimation and parametric analysis   |   |
| b. Optimization and trade studies  |   |
| <b>Week-11</b>   | <b>DESIGN CASE STUDY-I</b>                    |
| a. Design study of DC-3  |   |
| b. Design study B-747  |   |
| <b>Week-12</b>   | <b>DESIGN CASE STUDY-II</b>                   |
| I. Dynamics of F-16  |   |
| II. Dynamics of SR-71  |   |
| <b>REFERENCES:</b>   |   |
| 1. Daniel P. Raymer "Aircraft design a conceptual approach", 5 <sup>th</sup> Edition 1999. |   |
| <b>Course Home Page:</b>   |   |
| <b>SOFTWARE AND HARDWARE REQUIREMENTS FOR A BATCH OF 36 STUDENTS:</b>                      |   |
| <b>SOFTWARE:</b> Microsoft office excel spread sheet, MATLAB, AutoCAD Tool.                |   |
| <b>HARDWARE:</b> Desktop Computers with 4 GB RAM 36 nos                                    |   |