SPACE PROPULSION

VI Semester: AE

Course Code	Category	Ho	urs / W	/eek	Credits	Max	imum N	larks
A A E012	Corro	L	Т	Р	С	CIA	SEE	Total
AALU12	Core	3	1	-	4	30	70	100
Contact Classes: 45	Tutorial Classes: 15	Practical Classes: Nil			Total Classes: 60			

I. COURSE OVERVIEW:

An aerospace propulsion system is a device that generates forces to push the aerospace vehicles forward. This course discusses about the various Aerospace propulsive devices in micro level, it includes an overview of different types of propulsive system present in aircraftsand rockets such as turbojet, turboprop, turbofan, IC engine, solid propellant, hybrid propellant and liquid propellant engines. Along with that design and analysis will be discussed on the various parameters and components present in aerospace propulsive system.

II. OBJECTIVES:

The course should enable the students to:

- I. Evaluate various space missions, parameters to be considered for designing trajectories and rocket mission profiles
- II. Understand the fundamentals of chemical rocket propulsion, types of igniters and performance considerations of rockets.
- III. Discuss the working principle of solid and liquid propellant rockets and gain basic knowledge of hybrid rocket propulsion.
- IV. Illustrate electric propulsion techniques, ion and nuclear rocket and the performances of different advanced propulsion systems.

III. COURES OUTCOMES:

After successful completion of the course, students should be able to:

- CO 1 **Explain** the different types of air-breathing engines using the by using Brayton Understand cycle for selecting the suitable engine to an aircraft.
- CO 2 **Determine** the performance parameters that will influence the Air-breathing Apply engines by using parameters like SFC, Thrust, Efficienciesequation for improving the range and endurance of the aircraft.
- CO 3 **Select** the functions and geometrical parameters of different types of inlets, nozzles, Apply combustors and afterburners using aerodynamic principles for selecting of suitable design as per the engine specification.
- CO 4 Make a use of velocity triangle method to simplify the performance instability of the Apply compressor and turbine blades for increasing the efficiency of the engine.
- CO 5 Utilize the working principles of solid and hybrid rocket motors using Newton's Apply second law of equilibrium for simplifying the limitations and performances level of rocket engine.
- CO 6 **Distinguish** the different propellant feed system and injectors by using equations for Analyze thrust vector control for selecting the suitable component in rocket engine.

IV. SYLLABUS:

UNIT-I PRINCIPLES OF ROCKET PROPULSION

Classes: 09

History of rockets, Newtons third law, orbits and space flight, types of orbits, basic orbital equations, elliptical transfer orbits, launch trajectories, the velocity increment needed for launch, the thermal rocket engine, concepts of vertical takeoff and landing, SSTO and TSTO, launch assists.

-	UNIT-II	FUNDAMENTALS OF ROCKET PROPULSION	Classes: 09			
	Operating classificati nozzle les instrument	principle, Rocket equation, Specific impulse of a rocket, internal ballistics, l on, Rocket performance considerations of rockets, types of igniters, prelimina- is propulsion, air augmented rockets, pulse rocket motors, static testing of ation, safety considerations.	Rocket nozzle ry concepts in f rockets and			
	UNIT-III	SOLID ROCKET PROPULSION	Classes: 09			
]	Salient fea propellant a Erosive bu	tures of solid propellant rockets, selection criteria of solid propellants, estim adiabatic flame temperature, propellant grain design considerations. urning in solid propellant rockets, combustion instability, strand burner a	ation of solid			
	application	s and advantages of solid propellant rockets.				
	UNIT-IV	LIQUID AND HYBRID ROCKET PROPULSION	Classes: 09			
	Salient fea injectors for associated associated reverse hyb	tures of liquid propellant rockets, selection of liquid propellants, various feed or liquid propellant rockets, thrust control cooling in liquid propellant rock heat transfer problems, combustion instability in liquid propellant rockets, peed with operation of cryogenic engines, introduction to hybrid rocket propulsion, orid systems, combustion mechanism in hybrid propellant rockets, applications ar	l systems and ckets and the iliar problems standard and id limitations.			
	UNIT-V	ADVANCED PROPULSION TECHNIQUES	Classes: 09			
•	Electric rocket propulsion, types of electric propulsion techniques, Ion propulsion, Nuclear rocket comparison of performance of these propulsion systems with chemical rocket propulsion systems, future applications of electric propulsion systems, Solar sail.					
1	Text Book	s:				
	 Hill, P. Wesley Turner, Hieter 	G. and Peterson, C.R., "Mechanics and Thermodynamics of Propulsion", 2 nd Ed 7, 1992. , M.J.L., "Rocket and Spacecraft Propulsion", 2 nd Edition, MIT Press, 1922. and Pratt, "Hypersonic Air breathing propulsion" 5 th Edition, 1993.	ition, Addison			
	Reference	Books:				
	 Sutton, Mathur Distrib Tajmar 	G.P., "Rocket Propulison Elements" John Wiley & Sons Inc., New York, 5 th Edi r, M.L., and Sharma, R.P., "Gas Turbine, Jet and Rocket Propulsion", Standard I utors, Delhi, 1988. r, M., Advanced Space Propulsion Systems, Springer 2003.	tion, 1993. Publishers and			
	Web Refer	rences:				
	 https:// https:// https:// 	nptel.ac.in/courses/101106033/ nptel.ac.in/courses/112106073/ www.coursera.org/specializations/propulsion.				
	E-Text Bo	oks:				
	 https:// https:// https:// https:// https:// 	www.scribd.com/document/63588270/Aerospace-Propulsion-Systems as.wiley.com/WileyCDA/WileyTitle/productCd-1118806778.html as.wiley.com/WileyCDA/WileyTitle/productCd-1118307984.html as.wiley.com/WileyCDA/WileyTitle/productCd-0470824972.html				
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