# AEROSPACE STRUCTURAL DYNAMICS

VII SEMESTER: AE								
<b>Course Code</b>	Category	Hours / Week Credits			Maximum Marks			
AAEB25	Core	L	Т	Р	С	CIA	SEE	Total
		3	-	-	3	30	70	100
Contact Classes: 45	Tutorial Classes: Nil	Practical Classes: Nil			Total Classes: 45			

### I. COURSE OVERVIEW:

The course aim is to teach basic concepts and recent developments related to mechanical vibrations, structural dynamics and vibration control. The course seeks to introduce students to the fundamentals of dynamics by providing an overview on mechanical vibration. Vibrations in machines and structures are typically undesirable as they produce stresses, energy losses and increased bearing loads. They contribute to structural wear and can lead to passenger discomfort in vehicles. This course covers the vibrations of discrete systems and continuous structures and introduces the computational dynamics of linear engineering systems. Learn how to derive equations of motion and design vibration isolation systems. Gain an understanding of the concepts of natural frequencies and mode shapes and their significance. Complete system modeling tasks and formulate equations to measure and ultimately minimize vibrations. The concepts of aero elasticity phenomena, effect of aero elasticity in flight vehicle design.

#### **II. OBJECTIVES:**

#### The course should enable the students to:

- I. Demonstrate the knowledge of mathematics, science, and engineering by developing the equations of motion for vibratory systems and solving for the free and forced response.
- II. Understand to identify, formulate and solve engineering problems. This will be accomplished by having students' model, analyze and modify a vibratory structure order to achieve specified requirements.
- III. Introduce to structural vibrations which may affect safety and reliability of engineering systems.
- IV. Describe structural dynamic and steady and unsteady aerodynamics aspects of airframe and its components of space structures.

### **III. COURSE OUTCOMES:**

- CO 1 Explain the concepts of the equation of motion of free vibration and its Understand response for determining the nature of single degree of freedom.
- CO 2 **Demonstrate** the response of step function, periodic excitation (Fourier Understand series and transform, Laplace transform) of Single DOF for determining the freely vibrating of a body.
- CO 3 **Construct** the equation of motion of free vibration for the design of the Apply analysis of the spring-mass system.
- CO 4 Apply the various equations of forced vibration for determining the Apply frequency of the body.
- CO 5 Understand the torsional vibrations of rotor and geared systems for Understand determining the DOF of the vibrating systems.
- CO 6 **Develop** the formulation of stiffness and flexibility influence coefficients Apply for simplifying solution of multi DOF systems.

•	<b>e</b> the transverse, longitudinal, torsional and lateral vibrations of rods and beams for the design of continue elastic body.	Analyze
	stand the difference between the static and dynamic aeroelasticity I ermining the aeroelastic model of airfoils.	Understand
wing s	the static and dynamic aeroelasticity of the typical airfoil and ections of aircraft using Eigen functions and Laplace equation for of aircraft wing.	Analyze
IV. SYLLABUS	-	
MODULE-I	SINGLE-DEGREE-OF-FREEDOM LINEAR SYSTEMS	Classes: 10
response to an in	heory of vibration, equation of motion, free vibration, response to harmon pulsive excitation, response to a step excitation, response to periodic exci- to a periodic excitation (Fourier transform), Laplace transform (Transfer Fu-	tation (Fourier
MODULE-II	TWO-DEGREE-OF-FREEDOM SYSTEMS	Classes: 10
System, Torsiona Semi definite Sy	uations of Motion for Forced Vibration, Free Vibration Analysis of a al System, Coordinate Coupling and Principal Coordinates, Forced-Vibra rstems, Self-Excitation and Stability Analysis, Transfer- Function Appro ransform, Solutions Using Frequency Transfer Functions.	tion Analysis,
MODULE-III	MULTI-DEGREE-OF-FREEDOM LINEAR SYSTEMS	Classes: 08
and their propert	on, stiffness and flexibility influence coefficients; Eigen value problem; ies; Free and forced vibration by Modal analysis; Method of matrix invers ti- rotor systems and geared systems; Discrete- Time systems.	
MODULE-IV	DYNAMICS OF CONTINUOUS ELASTIC BODIES	Classes: 09
	hsverse vibration of a string or cable, longitudinal vibration of a bar or or rod, lateral vibration of beams, the Rayleigh-Ritz method.	rod, torsional
MODULE-V	INTRODUCTION TO AEROELASTICITY	Classes: 08
Typical Section Representation of Method. <b>Dynam</b>	<b>icity;</b> Typical Section Model of an Airfoil: Typical Section Model with Co Model—Nonlinear Effects. One Dimensional Aeroelastic Model of Airfor of Large Aspect Ratio Wing, Eigenvalue and Eigen function Approac <b>ic Aeroelasticity;</b> Hamilton's Principle: Single Particle, Many Particle Energy, Non potential Forces, Lagrange's Equations.	ils: Beam-Rod ch, Galerkin's
V. Text Books:		
	sr, M.N., "Structural Dynamics in Aeronautical Engineering", AIAA Educ	ation Series,
	999. Iechanical Vibrations", Prentice-Hall, 5 <sup>th</sup> Edition, 2011. ell, "A Modern Course in Aeroelasticity" Volume 217, Duke University,	Durham, NC,

#### VI. Reference Books:

- R.L. Bisplinghoff, H.Ashley, and R.L. Halfmann, "Aeroelasticity", Addison Wesley Publishing Co., Inc., 2<sup>nd</sup> Edition, 1996.
- 2. Leissa, A.W., Vibration of continuous system, The McGraw-Hill Company, 2<sup>nd</sup> Edition, 2011.
- 3. Inman, D.J., Vibration Engineering, Prentice Hall Int., Inc., 3<sup>rd</sup> Edition, 2001.

## VII. Web References:

- 1. http://ase.sbu.ac.ir/FA/Staff/abbasrahi/Lists/Dars/Attachments/11/Vibrations%20of%20Continuous% 20Systems.pdf
- 2. http://arc-test.aiaa.org/doi/book/10.2514/4.862458
- 3. http://arc-test.aiaa.org/doi/abs/10.2514/5.9781600862373.0719.0728

# VIII. E-Text Books:

- 1. http://www.gregorypaulblog.com/structural-dynamics-in-aeronautical-engineering-aiaa-education-series.pdf
- 2. https://aerocastle.files.wordpress.com/2012/10/mechanical\_vibrations\_5th-edition\_s-s-rao.pdf